

# SNS COLLEGE OF TECHNOLOGY



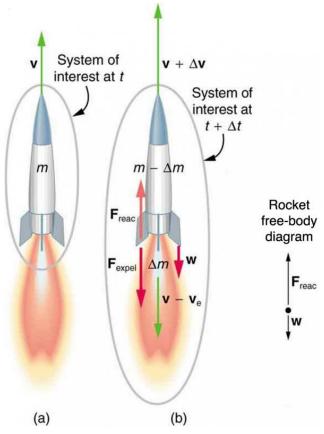
(An Autonomous Institution)
DEPARTMENT OF AEROSPACE ENGINEERING

Subject Code & Name: 23AST101 Fundamentals of Aerospace Engineering

#### **Topic: Principles of operation of rocket**

## Principles of Operation of a Rocket

Rockets operate on the principle of Newton's third law of motion: for every action, there is an equal and opposite reaction. The rocket expels exhaust gases at high speed in one direction, which generates thrust in the opposite direction, propelling the rocket forward. Here's a detailed look at the principles of operation of a rocket:



# 1. Basic Components of a Rocket

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# Principles of Operation of a Rocket

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#### **Basic Components of a Rocket**

1. Propellant:

- **Function:** The fuel and oxidizer that undergo combustion to produce high-pressure, high-velocity gases.
- **Types:** 
  - Liquid Propellant: Separate tanks for fuel (like liquid hydrogen) and oxidizer (like liquid oxygen).
  - Solid Propellant: A mixture of fuel and oxidizer in a solid form.
  - **Hybrid Propellant:** Combines elements of both liquid and solid propellants.

#### 2. Combustion Chamber:

- **Function:** The space where the fuel and oxidizer combust to generate high-pressure gases.
- **Operation:** The propellant is ignited in the combustion chamber, creating high-temperature, high-pressure gases.

#### 3. Nozzle:

- **Function:** Converts the high-pressure gases into a high-velocity exhaust jet.
- **Components:** 
  - **Convergent Section:** Narrows down to increase gas velocity.
  - **Throat:** The narrowest part where the gas velocity reaches sonic speed.
  - **Divergent Section:** Expands to further accelerate the gases to supersonic speeds.

#### 4. Structure:

- **Function:** The body of the rocket that houses all components.
- **Components:** Made of lightweight and strong materials to withstand the stresses of launch and flight.

#### **Principles of Rocket Operation**

#### 1. Propellant Storage:

- Liquid propellants are stored in separate tanks and are pumped into the combustion chamber.
- Solid propellants are pre-mixed and stored directly in the combustion chamber.

#### 2. Ignition:

• **Function:** Initiates the combustion process.

Methods: Spark ignition, hypergolic ignition (spontaneous ignition upon contact), or other methods depending on the propellant type.

#### 3. Combustion:

- The fuel and oxidizer mix and burn in the combustion chamber, producing high-pressure, high-temperature gases.
- $\circ$   $\;$  The rapid expansion of these gases creates thrust.

## 4. Thrust Generation:

- Principle: According to Newton's third law, the expulsion of gases at high speed through the nozzle produces an equal and opposite force that propels the rocket forward.
- **Equation:**  $F=m \cdot veF = \int dot\{m\} \int eF=m \cdot ve$ 
  - FFF = Thrust
  - m \dot{m}m = Mass flow rate of exhaust gases
  - vev\_eve = Exhaust velocity

## 5. Nozzle Function:

- **Convergent Section:** Compresses and accelerates the gases.
- **Throat:** The point where gases reach sonic speeds.
- **Divergent Section:** Further accelerates the gases to supersonic speeds, maximizing thrust.
- 6. Staging:
  - **Function:** Used in multi-stage rockets to discard parts of the rocket that are no longer needed, reducing weight and increasing efficiency.
  - **Operation:** As the lower stage burns out, it is jettisoned, and the next stage ignites.

#### **Advanced Concepts**

# 1. Specific Impulse (Isp):

- **Definition:** A measure of the efficiency of a rocket engine, defined as the thrust per unit weight flow rate of the propellant.
- **Equation:**  $Isp=Fm \cdot g0I_{sp} = \frac{F}{dot_m} \quad g0F$ 
  - FFF = Thrust
  - m<sup>`</sup>\dot{m}m<sup>`</sup> = Mass flow rate of propellant
  - g0g\_0g0 = Standard acceleration due to gravity
- 2. Delta-V (Δv):

- Definition: The change in velocity that a rocket can achieve, important for mission planning and determining the capability to reach desired orbits or destinations.
- - m0m\_0m0 = Initial mass
  - mfm\_fmf = Final mass

## 3. Thrust-to-Weight Ratio:

- **Definition:** The ratio of thrust produced by the rocket to its weight.
- **Importance:** Determines the rocket's ability to lift off and accelerate.
- Equation: Thrust-to-Weight Ratio=Fm·g0\text{Thrust-to-Weight Ratio} = \frac{F}{m \cdot g\_0}Thrust-to-Weight Ratio=m·g0F

## **Applications of Rockets**

## 1. Space Exploration:

- Launching satellites, space probes, and manned spacecraft into orbit and beyond.
- Examples: Saturn V (Apollo missions), Falcon 9 (SpaceX), Ariane 5 (ESA).

## 2. Military:

- Missiles and strategic weapons.
- Examples: Intercontinental Ballistic Missiles (ICBMs), Cruise Missiles.

# 3. Scientific Research:

- Launching scientific instruments and experiments into space.
- Examples: Sounding rockets for atmospheric research, space telescopes like Hubble.

# 4. Commercial:

- Commercial satellite launches, space tourism.
- Examples: Communication satellites, space tourism ventures like Blue
   Origin and Virgin Galactic.

# 5. Environmental Monitoring:

- Earth observation satellites for climate monitoring, disaster management.
- Examples: NOAA weather satellites, Landsat program.

#### Summary

Rockets operate on fundamental physical principles, primarily Newton's third law of motion, to generate thrust and propel vehicles through the atmosphere and into space. Understanding the components and operation of rockets is crucial for appreciating their role in modern technology, space exploration, defense, and various commercial and scientific applications.