

## SNS COLLEGE OF TECHNOLOGY (An Autonomous Institution)



## DEPARTMENT OF AERONAUTICAL ENGINEERING

Subject Code & Name: 19AST203 Aircraft Structural Mechanics

TOPIC: Bredt – Batho formula

## **Bredt-Batho formulae**

The 1st Bredt-Batho formula indicates the relationship between the torsional moment MT acting on a thin-walled hollow tube, its enclosed area Am and the resultant shear flow T / shear stress  $\tau$ . It is calculated as follows:  $\tau$ =Tt=MT2·Am·t.

The variable t represents the thin-walled component's wall thickness. The enclosed area Am lies within the centre line of the tube and is also called the hollow area. The shear stress  $\tau$  resulting from the <u>Torsion</u> is constant over the entire wall thickness t, which means that the shear flow T also remains constant in the circumferential direction.

The 2nd Bredt-Batho formula indicates the component's twisting  $\vartheta$ , which depends on the material's shear modulus G. A component's torsional <u>Resistance</u> I<sub>T</sub> can also be determined.

The Bredt-Bredt-Batho formulae apply only to torsion acting on closed hollow tubes with an axis of <u>Rotation</u> that lies on the shear centre.

A beam with a closed section experiencing only a pure torque T and without any axial constraints, does not develop direct stresses, ie s z = 0.

So equations (4.2) and (4.3) become:

$$\frac{\partial q}{\partial s} = \frac{\partial q}{\partial z} = 0$$

The only way to satisfy these equations would be if the shear flow 'q' was constant.

NOTE: Although 'q' is constant, the shear stress 't ' may not be if the wall thickness 't' varied with 's'.

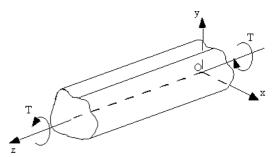


Figure 45: Closed beam with applied torque.

To determine the relationship between applied torque and shear flow,

apply equilibrium to the end of the beam.

In essence the applied Torque T must equal to the torque generated by the shear flow.

Look at the end of beam, and a small section ds.

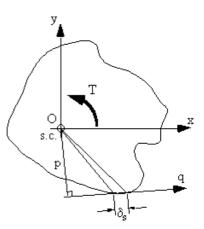


Figure 46: Equating applied torque with moment generated by shear flow.

The torque produced by the shear flow on element ds is pqsd s. Integrating about the whole section gives:

 $T = \oint pqds$ 

We have previously defined that:

$$\oint pds = 2A$$

Therefore:

$$q = \frac{T}{2A}$$
 (5.1)

Often referred to as the 'Bredt-Batho Formula'. Substituting this equation into (4.21) gives the rate of twist due to the Torque 'T':

Let us suppose the origins where the shear flow has unknown values 9 s. or sin all avias of Then for closed socian along as being a soprinte Qs = - Sx [xtds - Sy Jytds+ Qs,0 95=96+950 9310 = - M Ms unbalanced Moment 9 b > basic Shear flow equation for open tube 95.0 > Unknown shearflow equation at origin of 's 'co-ordinates. The value of shear-flow at origin of 's' is found by making a cut at that point .. and equating applied the initial moments taken abo some convinience point. ) plot Shearflow for given closed Section × 160 1000N x = 18783 - 18 - 18 adl EA 40 0.0 2 H given Hem

To solve the above problem; make a curb, between Solure and wanted sailing Stringes a and d to make it as open section. pecomos - 215B NUT Longolding trama)1 Ay Ax2 Ayl Axy x y Ax Boom Area 0 3600 4 00 0 30 0 120 are H atoribra 03 2 20 H 0, 10 01 01 10 0190 50 6 and 2 to Ho 101 86 10 32001 00 84 000 à 2 40 20' 80 40 8000 K800 1900 50 160 160 6 HOO HAOO 1600 80 12 2 rop ware  $\frac{\epsilon_{Ax}}{\epsilon_{A}} = \frac{160}{12}$ x x = 13.33  $\frac{z_{Ay}}{z_{A}} = \frac{160}{12}$ 4 13.33 4 EIgx + EAY - EAY

$$F x x = h h 00 - (2(13.32)^{2})$$

$$F x x = 2267.744 cm^{4}$$

$$T y y = 5 y y + 5 A^{2} - 5 A^{2}$$

$$T y y = 6 h 00 - (2(13.33)^{2})$$

$$T y y = 4267.73 cm^{4}$$

$$T x y = 5 A x y - 5 A^{2} y$$

$$T x y = (600 - (2(13.33))(13.32))$$

$$T x y = -53 2.26 cm^{4}$$

$$S x = 0 - [000 \times (-532.26)^{2}]$$

$$T x x T y y$$

$$S x = 0 - [000 \times (-532.26)^{2}]$$

$$T x y = -53 x - 5 y$$

$$T x y = (-532.26)^{2}$$

$$S x = 2441.71 N$$

$$S y = \frac{1 - \frac{1}{2} x y}{T x x T y y}$$

$$S y = \frac{1000 - [0 \times (-532.26)^{2}]}{1 - \frac{1}{2} 2 x y}$$

$$S y = \frac{1000 - [0 \times (-532.26)^{2}]}{(-532.26)^{2}}$$

9=-Sy ERiyi - Sr Izz ERiyi - Iyy UND HOULDES -9 = 10 - 1030.18 EATYI - 241.71, EATXI LOT 100 - 6 000 - 12 (13. 23) 9=-0.45 SAIYi - 0.05 SAIXi TANKS -TYDE FAXU D B H1 2 . 21 01 -0001 - 10x1 H -13.33-13.33 26.67 26.67 x 6.67 16.67 -13.33 -13.32 9ab=-0.45 (4)(16.67)-0.05 (4) (-13.33) 9ab = -30.27 + 3.017 - 500- - 0 9ab = -27.26 Scal. 083-2 9 bc = - 0.45(4)(-13.33) - 0:05(4)(-13.33) + 9ab 960 = 24.20 + 3.017 - 27.26 11 11 - 1HC = 22 9bc = 0 NXT 9cd=-0.45(2)(-13.83)-0.05(2)(26.67)+96c PAXE 9cd = 12.10 - 3.019+0. UNEXCE (00.022-1) 9 cd = 9.05 N (cm NP. Fare \* 0 0001 · (430.082-) 1000N 14 10 1. 1000 N 81.00 M

$$\begin{aligned} q_{s} = q_{b} + q_{s,b} \\ q_{s,b} = -\frac{M}{2A} \\ A = \frac{1}{2} \pi r^{2} + \frac{1}{3} bb \\ A = \left(\frac{1}{2} \times \pi \times 1s^{2} + \frac{1}{2} \times (\mu \circ \times 1b) + (\mu \circ \times 20)\right) \\ A = (353.42 \text{ cm}^{2} \\ \text{Moment about b}^{2} b^{2} (\text{unbalanced toment)} \\ A = (-\mu \circ \circ \times 10 - 2) \left(\frac{\pi r^{2}}{2}\right) \times 27.26\right) \\ + q_{0} \circ s \times Ho \times 20 \\ = (0000 - 2\pi \cdot 26 \times \pi \times 15^{2} + q_{0} \circ 8 \times 800) \\ = -2\pi 36 - 2\pi \cdot 26 \times \pi \times 15^{2} \\ \text{Moment about for a both of the second se$$

9bc = 0+8.13 9bc = 8.13 N lcm 9cd = 9.08+8.13 - C- 22004 .95) - 1° ett 2cd = 17.21 NICM 2 × 1353.45 9510 = -25.0 = 8.13 N Com 95,0 mo ch. d' Just D AUN)8.12 17.21 13 C=1× TFX 30. FC - 3 B-IN Torsional effect of Multicell tube: M= - 22004 . 95 N 10m Assumption: -1. Angle of twist is equalifor all cells?  $\Theta_1 = \Theta_2 = \Theta_3 = \Theta$ hop - all 2. Material is homogeneous boros to walt 1125 \$. It obeys hooks law. H. Beam is subjected to torque alo alor F1. P A 20, De for single cell 2ab = -19.13 T= 2A9

